

## LECTURE 9: LINEAR ELASTICITY...CONTINUED

### 6.5 Incompressible linear elasticity

Recall: Incompressibility:

$$\det \mathbf{F} = 1, \quad \implies \det(\mathbf{1} + \mathbf{H}) = 1 + \text{tr } \mathbf{H} + \mathcal{O}(\mathbf{H}^2) = 1$$

Therefore  $\text{tr } \mathbf{H} = 0 = \text{Div } \mathbf{u}$ , and

$$\boxed{\text{Div } \mathbf{u} = 0} \iff \boxed{\text{tr } \mathbf{E} = 0}$$

Also

$$\mathbf{T} = -p\mathbf{1} + J^{-1}\mathbf{F}\frac{\partial W}{\partial \mathbf{F}}, \quad \sigma = -p\mathbf{1} + C_{ijkl}e_{kl}$$

For isotropic material,

$$\sigma = 2\mu\mathbf{E} + \lambda(\text{tr } \mathbf{E})\mathbf{1} - p\mathbf{1}$$

but

$$\mu = \frac{E}{2(1+\nu)} = \frac{E}{3}.$$

Therefore

$$\boxed{\rho\ddot{\mathbf{u}} = \mathbf{b} - \text{Grad } p + \mu\Delta\mathbf{u}}$$

and

$$\boxed{\mu = \frac{E}{3}}$$

*N.B.* Boundary conditions

$$\begin{aligned} \mathbf{u} &= \mathbf{u}^*(t), & \text{on } \partial_1\mathcal{B} & \text{ displacements} \\ \boldsymbol{\tau}\mathbf{n} &= \mathbf{t}^*(t), & \text{on } \partial_2\mathcal{B} & \text{ traction} \end{aligned}$$

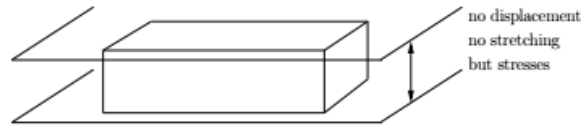
#### 6.5.1 General principles

- 1) Linear superposition
- 2) Stresses, strains and displacements are proportional to the loads (or displacements) applied to the solid.
- 3) If  $\partial_2\mathcal{B} = \emptyset$  then there exists one unique solution, only displacements.
- 4) If only traction and tractions are in equilibrium, then stresses and strains are unique. For initial conditions, there exists one unique  $u(t)$ .

Some nomenclature about loading

- 1) Plain strain

$$\mathbf{u} = (u(X, Y), v(X, Y), 0) \implies e_{13} = e_{23} = e_{33} = 0, \quad \tau_{13} = \tau_{23} = \tau_{31} = \tau_{32} = 0.$$



2) Plane stress

$$\tau_{13} = \tau_{23} = \tau_{33} = 0, \quad \tau = \begin{bmatrix} \text{ast} & \text{ast} & 0 \\ \text{ast} & \text{ast} & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

3) Antiplane strain

$$\mathbf{u} = (0, 0, w(X, Y))$$

4) Pure torsion

$$\mathbf{u} = (-\Omega Y Z, \Omega X Z, \Omega \varphi(X, Y))$$

(see problem sheet 6)

## 6.6 Plane/Strain/Stress Solutions

### 6.6.1 Plane solutions

(stress or strains)

In cartesian coordinates,

$$e_{ij} = \frac{1}{2}(u_{i,j} + u_{j,i}) = \frac{1+\nu}{E}\tau_{ij} - \frac{\nu}{E}\tau_{kk}\delta_{ij}$$

$$\frac{\partial \tau_{ij}}{\partial x_i} + b_j = 0$$

Assume that  $b$  derives from a potential,

$$b_i = \frac{\partial V}{\partial X_i}, \quad i = 1, 2, \quad b_3 = 0.$$

Plane stresses or strain  $\tau_{13} = \tau_{23} = 0$ .

### 6.6.2 Idea

Let

$$\tau_{11} = \frac{\partial \phi}{\partial X_2} - V, \quad \tau_{22} = \frac{\partial \phi}{\partial X_1} - V, \quad \tau_{12} = -\frac{\partial^2 \phi}{\partial X_1 \partial X_2}, \quad \tau_{33} = \beta \nu (\tau_{11} + \tau_{22}),$$

$\beta = 0$  is plane stress and  $\beta = 1$  is plane strain.

6.6.3 Equations

$$\frac{\partial \tau_{11}}{\partial X_1} + \frac{\partial \tau_{12}}{\partial X_2} + b_1 = 0, \quad \frac{\partial \tau_{12}}{\partial X_1} + \frac{\partial \tau_{22}}{\partial X_2} + b_2 = 0.$$

Therefore

$$\begin{aligned} \frac{\partial}{\partial X_1} \left( \frac{\partial^2 \phi}{\partial X_2^2} - \mathcal{Y} \right) + \frac{\partial}{\partial X_2} \left( -\frac{\partial^2 \phi}{\partial X_1 \partial X_2} \right) + b_1 &= 0, \\ \frac{\partial}{\partial X_1} \left( -\frac{\partial^2 \phi}{\partial X_1 \partial X_2} \right) + \frac{\partial}{\partial X_2} \left( \frac{\partial^2 \phi}{\partial X_2^2} - \mathcal{Y} \right) + b_2 &= 0, \end{aligned}$$

and the equations of motion are satisfied. But we do not have an equation for  $\phi$ . We have equations for  $\tau_{ij}$  or  $e_{ij}$ , that is, 6 fields but  $u_i$  is 3 components.

6.6.4 Compatibility conditions

Recall: conditions for  $\mathbf{F}$ :  $\text{Curl } \mathbf{F} = 0$ . For

$$e_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

Compatibility conditions:

$$\text{Curl Curl } \mathbf{E} = 0,$$

$$\iff \epsilon_{ipm} \epsilon_{jqn} \frac{\partial^2 e_{mn}}{\partial X_p \partial X_q} = 0$$

$$\iff \frac{\partial^2 e_{ij}}{\partial X_k \partial X_l} + \frac{\partial^2 e_{kl}}{\partial X_i \partial X_j} - \frac{\partial^2 e_{il}}{\partial X_j \partial X_k} - \frac{\partial^2 e_{jk}}{\partial X_i \partial X_l} = 0$$

These are 6 relations (but only 3 are independent). For planar problems:  $e_{13} = e_{23} = 0$ ,  $\partial e_{ij} / \partial X_3 = 0$ ,

$$\implies \frac{\partial^2 e_{11}}{\partial X_2^2} + \frac{\partial^2 e_{22}}{\partial X_1^2} - 2 \frac{\partial^2 e_{12}}{\partial X_1 \partial X_2} = 0 \quad (\text{ast})$$

Now for plane stress we have  $\tau_{33} = 0$  and from plane strain we have  $\tau_{33} = \nu(\tau_{11} + \tau_{22})$ ,

$$\iff \tau_{33} = \beta \nu (\tau_{11} + \tau_{22}),$$

which implies

$$\begin{aligned} e_{11} &= \frac{1+\nu}{E} \tau_{11} - \frac{\nu}{E} (1+\beta \nu) (\tau_{11} + \tau_{22}) \\ e_{22} &= \frac{1+\nu}{E} \tau_{22} - \frac{\nu}{E} (1+\beta \nu) (\tau_{11} + \tau_{22}) \\ e_{12} &= \frac{1+\nu}{E} \tau_{12} \end{aligned}$$

Insert these into (ast) and use  $\tau_{11} = \frac{\partial^2 \phi}{\partial x_1^2} - V$ ,

$$\begin{aligned} \Rightarrow \frac{\partial^4 \phi}{\partial x_1^4} + 2 \frac{\partial^4 \phi}{\partial x_1^2 \partial x_2^2} + \frac{\partial^4 \phi}{\partial x_2^4} &= \frac{1 - \beta \nu^2}{1 - \nu - 2\beta \nu^2} \left( \frac{\partial^2 V}{\partial x_1^2} + \frac{\partial^2 V}{\partial x_2^2} \right) \\ \Leftrightarrow \boxed{\nabla^4 \phi = C_\nu \Delta V}, \quad C_\nu &= \frac{1 - \beta \nu^2}{1 - \nu - 2\beta \nu^2}. \end{aligned}$$

Here  $\nabla^4$  is the *biharmonic operator* and  $\phi$  is the *Airy potential*. If  $\beta = 0$ , we have plane stress and  $\beta = 1$  is plane strain.

### 6.6.5 Application

## 6.7 Elasto-dynamics

$$\mu \Delta \mathbf{u} + (\lambda + \mu) \text{Grad Div } \mathbf{u} = \rho \ddot{\mathbf{u}} \quad (\text{ast})$$

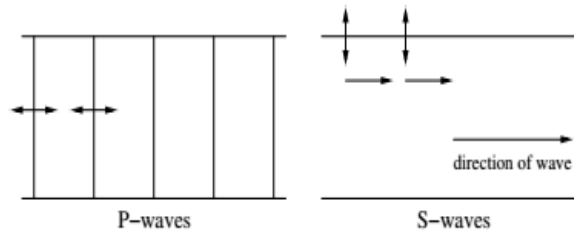
### 6.7.1 Planar waves

$$\mathbf{u}(\mathbf{x}, t) = \mathbf{a} \sin(\mathbf{k} \cdot \mathbf{x} - ct)$$

Here  $\mathbf{a}$  is the amplitude,  $\mathbf{k}$  is the direction and  $c$  is the velocity. We normalize such that  $|\mathbf{k}| = 1$ .

2 interesting cases:

- $\mathbf{a} \parallel \mathbf{k}$  longitudinal – primary, pressure, P-waves.
- $\mathbf{a} \perp \mathbf{k}$  transverse – shear, secondary, S-waves.



Let  $\varphi(\mathbf{x}, t) = \mathbf{k} \cdot \mathbf{x} - ct$ .

Note that  $\text{Div } \mathbf{u} = \mathbf{a} \cdot \mathbf{k} \cos \varphi$

$\text{Curl } \mathbf{u} = -\mathbf{a} \times \mathbf{k} \cos \varphi$

$\text{Div } \mathbf{u} = 0$  is transverse,  $\text{Curl } \mathbf{u} = 0$  is longitudinal.

Substitute  $\mathbf{u} = \mathbf{a} \sin \varphi$  in (ast). Then

$$\Delta \mathbf{u} = -\mathbf{a} \sin \varphi$$

$$\text{Grad Div } \mathbf{u} = \text{Grad}(\mathbf{a} \cdot \mathbf{k} \cos \varphi) = (\mathbf{a} \cdot \mathbf{k}) \mathbf{k} (-\sin \varphi)$$

Therefore  $\ddot{\mathbf{u}} = -c^2 \mathbf{a} \sin \varphi$  and

$$\mu \mathbf{a} + (\lambda + \mu)(\mathbf{a} \cdot \mathbf{k}) \mathbf{k} = \rho c^2 \mathbf{a}$$

This is a linear operator on  $\mathbf{a}$ . Define  $A$  the *acoustic tensor*,

$$A = \frac{1}{\rho} (\mu \mathbf{1} + (\lambda + \mu) \mathbf{k} \otimes \mathbf{k}) [A]_{ij} = \frac{1}{\rho} (\mu \delta_{ij} + (\lambda + \mu) k_i k_j)$$

so that we have the eigenvalue problem

$$A\mathbf{a} = c^2 \mathbf{a}$$

1)  $\mathbf{a} = \alpha \mathbf{k}$

$$\alpha A_{ij} k_j = \frac{1}{\rho} (\mu k_i + (\lambda + \mu) \underbrace{k_j k_j}_{1} k_i) = c^2 \alpha k_i$$

$$\Rightarrow \frac{\lambda + 2\mu}{\rho} = c^2, \quad c_L = \sqrt{\frac{\lambda + 2\mu}{\rho}}$$

2)  $\mathbf{a} \perp \mathbf{k}, a_i k_i = 0$ .

$$A_{ij} a_j = \frac{1}{\rho} (\mu a_i + (\lambda + \mu) k_i k_j a_j) = c^2 a_i$$

$$\Rightarrow c^2 = \frac{\mu}{\rho}, \quad c_T = \sqrt{\frac{\mu}{\rho}} < \sqrt{\frac{\lambda + 2\mu}{\rho}},$$

*i.e.* slower than  $c_L$ .

Note also

$$c_L = \sqrt{\frac{E(1-\nu)}{(1+\nu)(1-2\nu)\rho}},$$

where  $1 - 2\nu = 0$  for an incompressible material. Therefore  $c_L \rightarrow \infty$  as  $\nu \rightarrow 1/2$ .

Also note

$$\begin{aligned} c_T^2 = \mu/\rho &\Rightarrow \mu = c_T^2 \rho \\ c_L^2 = \frac{\lambda}{\rho} + 2c_T^2 &\Rightarrow \rho c_L^2 - 2\rho c_T^2 \end{aligned}$$

$$\Rightarrow \boxed{\ddot{\mathbf{u}} = c_T^2 \Delta \mathbf{u} + (c_L^2 - c_T^2) \text{Grad Div } \mathbf{u}} \quad (\text{astast})$$

### 6.8 Rayleigh waves

$$\mathbf{u} = \Re \begin{bmatrix} A e^{-by} \exp(ik(x - ct)) \\ B e^{-by} \exp(ik(x - ct)) \\ 0 \end{bmatrix}$$

$A, B \in \mathbb{C}, b \in \mathbb{R}_0^+$ .

