
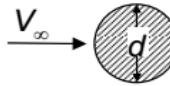
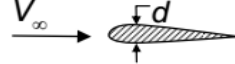
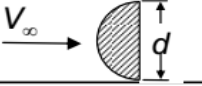
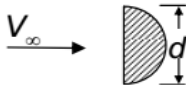
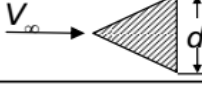
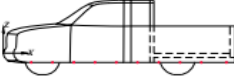

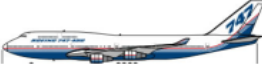


Aerodynamics: Drag

- The drag is an important subject in aerodynamics.
- A reduction in drag can lead to a reduction in fuel consumption and better performance for a vehicle.
- The drag coefficient varies from one object to another depending on the particular geometry of that object.
- For streamlined body such as wing and airfoil, the drag coefficient is low compared to bluff body such as circular cylinder, sphere or road vehicle.

Aerodynamics: Drag

Normal Plate		$C_D = 2.0$
Circular Cylinder		$C_D = 1.2$ at $Re = 10^5$ $C_D = 0.6$ at $Re = 10^7$
Streamlined body		$C_D = 0.12$
Half Cylinder		$C_D = 1.2$
Half Cylinder		$C_D = 1.7$
Equilateral triangle		$C_D = 1.6$
Pickup truck		$C_D = 0.4-0.5$
Piper PA-16 Clipper		$C_D = 0.037$
Boeing 747		$C_D = 0.017$

Aerodynamics: Drag for Airfoil vs. Wing

- It is important to note that there is a difference between the drag of an airfoil and that of a wing.
- The drag acting on an airfoil section is the sum of the *skin friction drag*, D_f , and the *pressure drag*, D_p , which is due to flow separation. That is,

$$C_d = \frac{D_f + D_p}{q_\infty S}$$

The sum of the skin friction drag and the pressure drag is called *profile drag*.

- On the other hand, the total drag of a subsonic finite wing in a real case is the sum of the induced drag, D_i , and the profile drag,

$$C_D = C_d + \frac{D_i}{q_\infty S}$$

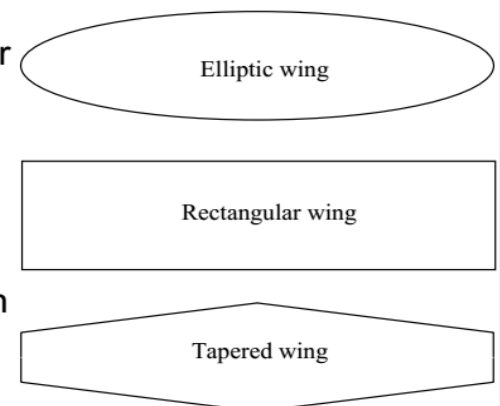
where the subscript D represents the drag of the wing and the subscript d represent the drag of the airfoil.

- Using the lifting line theory it can be shown that for a general wing

$$C_{D,i} = \frac{C_L^2}{\pi e \mathcal{R}}$$

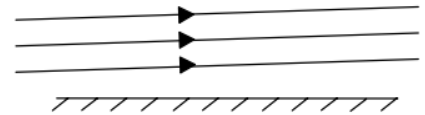
where $C_{D,i}$ is the induced drag coefficient and e is the *span efficiency factor*. For elliptical wing, $e = 1$ and for other platforms, $e < 1$.

- Therefore, the induced drag is minimum for an elliptical platform.
- In the past, several aircraft have been designed with elliptical wings.
- However, elliptical wings are more expensive to manufacture than other simple platform such as rectangular wings. The rectangular wing is considered far from optimum. A compromise between the elliptical wing (manufacturing difficulty) and rectangular wing (poor efficiency) is the tapered wing.

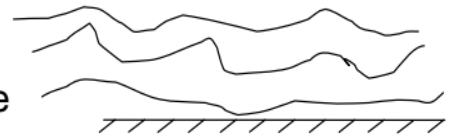


Aerodynamics: Laminar and Turbulent Flows

- The drag coefficient of a body depends on the flow around the body whether it is laminar or turbulent.
- When the streamlines are smooth and regular and a fluid element moves smoothly along a streamline the flow is called laminar.
- On the other hand, when the streamlines break up and a fluid element moves in a random, irregular, and tortuous fashion the flow is called turbulent.
- Most of real flows are turbulent flows.
- In turbulent flow, the higher energy fluid elements from the outer regions of the flow are pumped close to the surface. Hence, the average flow velocity near a solid surface is larger for a turbulent flow in comparison with laminar flow. Figure 15 shows the velocity profile for laminar and turbulent boundary layers.



Laminar flow



Turbulent flow

- In turbulent flow, the higher energy fluid elements from the outer regions of the flow are pumped close to the surface.
- Hence, the average flow velocity near a solid surface is larger for a turbulent flow in comparison with laminar flow.
- Since the shear stress is proportional to the velocity gradient along the y -direction $\tau \propto \partial u / \partial y$ then the shear stress (friction) as well as aerodynamic heating at the wall surface is higher for turbulent flow than laminar flow.

